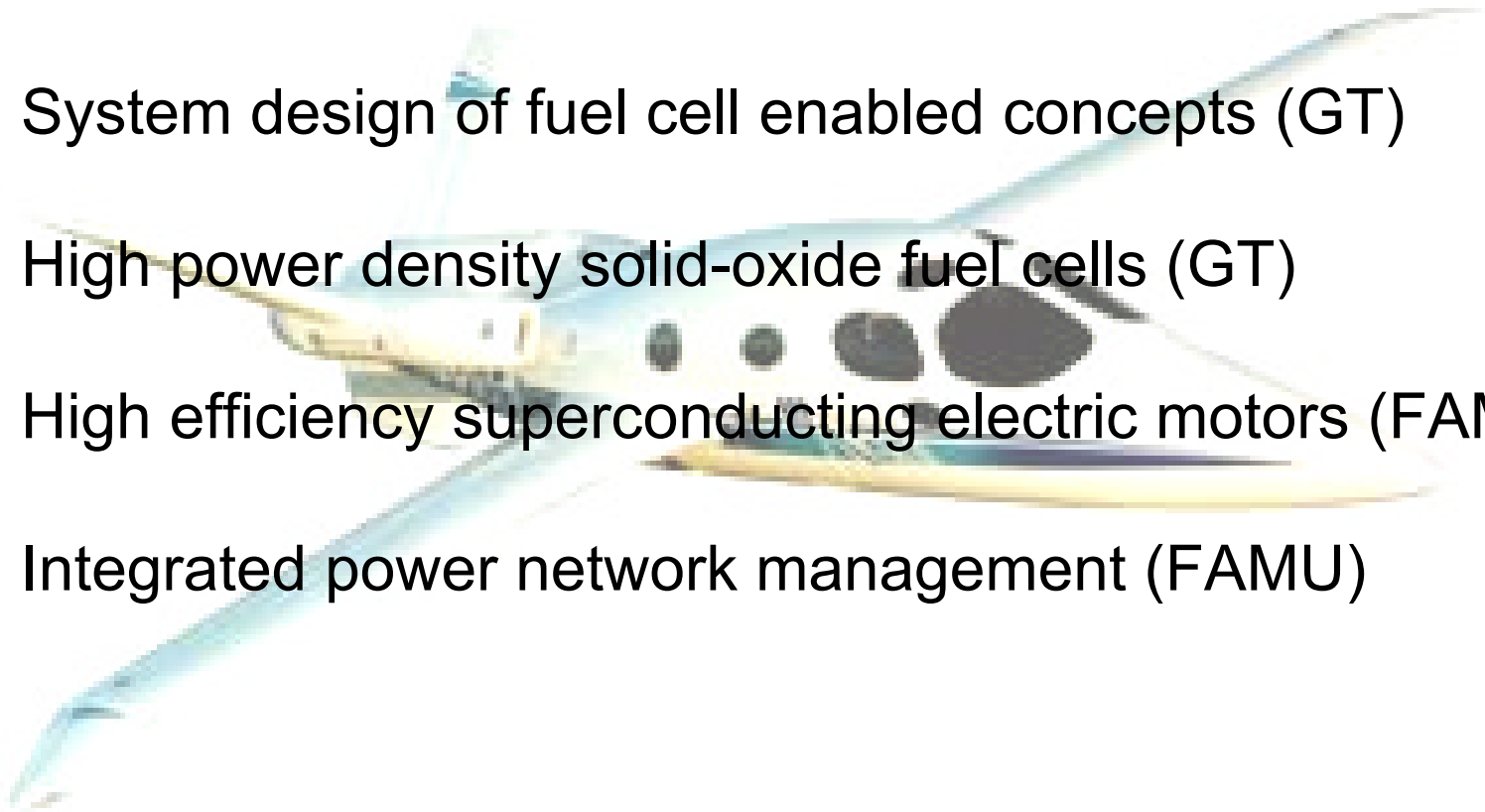


# Aeropower Thrust Area

- System design of fuel cell enabled concepts (GT)
- High power density solid-oxide fuel cells (GT)
- High efficiency superconducting electric motors (FAMU)
- Integrated power network management (FAMU)



# Automotive Example: GM AUTOnomy Concept



***“Fuel cells will power cars with little or no waste at all. We happen to believe that fuel cell cars are the wave of the future; that fuel cells offer incredible opportunity.”***

*President George W. Bush, February 25, 2002.*

## 2.5 Advanced Power Technology

### 2.5.1 Micro-structured Solid Oxide Fuel Cells for Aerospace Power Generation

David Parekh, Meilin Liu, Comas Haynes, Georgia Tech

#### Science & Technology Objective(s):

- Functionally graded electrodes for high power density
- Sulfur tolerant anodes suitable for use with jet fuels
- Modeling of steady and transient processes
- Solid oxide fuel cells for aircraft propulsion & APUs

#### Collaborations:

- Government - DOE, DARPA, NSF, ONR, ARO
- URETI - GT System Design & FAMU Power Network
- Industry - UTC Fuel Cells, Siemens, Delphi, Boeing
- Synergism with existing programs - DOE Solid-State Energy Conversion Alliance, DARPA Palm Power

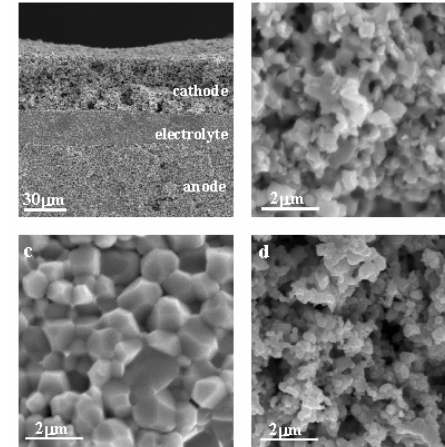
#### Proposed Approach:

- Development of new microstructured materials for high power density and sulfur tolerance
- Lagrangian modeling of electrochemical and thermodynamic processes
- System modeling and integration into demonstrator

#### NASA Relevance/Impact:

- Zero CO<sub>2</sub> emission, low-noise propulsion & power source
- Increased safety through distributed power
- All-electric, non-polluting civilian VSTOL transport

#### Microstructured materials enhance power density



#### Milestones/Accomplishments:

- Quantify requirements for power sources on current and future aircraft based on system studies
- Development of new materials specifically engineered to provide high power density required for flight applications
- Integrated power subsystem including fuel cell and superconducting motors
- Flight demonstration of unmanned, all-electric air vehicle powered by high power density solid oxide fuel cells

## 2.5.1 Proposed Approach

- **SYSTEM ANALYSIS:** Determine baseline requirements for propulsion and electrical power based on system analysis of revolutionary aircraft concepts.
- **MATERIALS:** Fabrication of electrodes and SOFCs using combustion CVD and microfabrication; characterization of fuel cell reactions using in-situ FTIR/Raman, impedance, and MS/GC measurements.
- **MODELING:** Mathematical modeling of functionally graded electrodes; Lagrangian electrochemical analysis of cell performance under steady and transient loads; first-order modeling of complete system including fuel processor, fuel cell stack, and power electronics.
- **INTEGRATION:** Develop fuel cell module ideally suited for power and weight constraints of flight application and characteristics of electrical architecture and loads.
- **DEMONSTRATION:** Design, develop, and fly subscale, unmanned prototype of fuel-cell powered aircraft utilizing high-density solid oxide fuel cells and unique electrical architecture and motors from Tasks 2.1.2 and 2.1.3.

# 2.5 Advanced Aeropower Technology

## 2.5.2 Integrated Power Management

Thomas Baldwin & Cesar Luongo, FAMU

### Science & Technology Objective(s):

- Examine the impact of distributed generation, automatic reconfiguration, power electronics, and super-conductivity on aircraft power systems
- Provide simulation capabilities to examine the performance of individual components and control strategies.

### Collaborations:

- Government - ONR, NASA LaRC (SAB)
- URETI - GT (ASDL), GTRI (Parekh)
- Industry – RTDS
- Synergism with existing programs – Navy All Electric Ship Program

### Proposed Approach:

- Examine alternative system configurations using real-time digital simulation coupled with hardware-in-the-loop modeling.
- Examine techniques for automatic system reconfiguration in response to failure or damage.
- Analyze the impact of superconducting and power electronic devices on power system performance.

### NASA Relevance/Impact:

- Increased reliability, safety and performance through improved redundancy and system reconfiguration
- Provide a test platform for testing new concepts and technologies in power distribution and controls

### Real-time simulation of a power network



### Milestones/Accomplishments:

- Model a base-line electric power system for the aircraft on the real-time digital simulator.
- Complete simulation studies of system reconfiguration and power distribution controls.
- Assessment of the impact of distributed generation and superconductivity on aircraft power systems.

## 2.5.2 Proposed Approach

- Examine alternative system configurations using real-time digital simulation (RTDS)
  - model a base-line electric power system for the specified aircraft on the real-time digital simulator
  - creating alternative designs of the power system and perform a comparison analysis (i.e. distributed generation and energy storage)
- Examine techniques for automatic system reconfiguration
  - model the automatic system reconfiguration and power controls on the RTDS simulator
  - simulate failures and damage to the aircraft power system and analyze the system responses and ability to restore power to critical flight systems
- Analyze the impact of new devices and technologies on performance
  - develop and insert RTDS simulator models for new technologies (i.e. super-conducting and power electronics devices)
  - perform a comparison analysis of the new technologies, including impact on weight and size as well as power distribution performance and control



# 2.5 Advanced Aeropower Technology

## 2.5.3 High Efficiency Superconducting Electric Motor

Cesar Luongo, FAMU

### Science & Technology Objective(s):

- Establish targets for aircraft-based s/c motors
- Develop s/c motor concepts to reach targets
- Integrate to other technologies for new concepts of all-electric aircraft

### Collaborations:

- Government - NASA GRC (ASAO), LaRC (SAB)
- URETI - GT (ASDL)
- Industry - American Superconductor, Boeing
- Synergism with existing programs - ONR all-electric ship (CAPS)

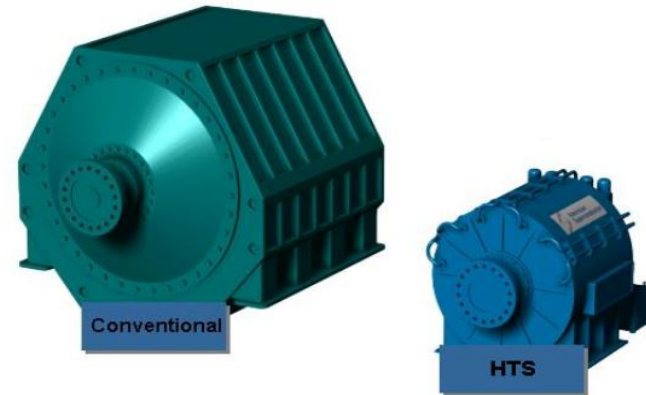
### Proposed Approach:

- Benchmark existing conventional and s/c motors
- Develop conceptual design of high power density s/c motor
- Integrate to all-electric aircraft concept

### NASA Relevance/Impact:

- Low weight/volume electric motors
- Increased use of electrical actuators, elimination of mechanical/hydraulic systems
- All-electric, non-polluting civilian transports

Space and weight reduction using superconductors



### Milestones/Accomplishments:

- Establish targets for material characteristics and s/c motor performance for air-worthiness, determine relevant ratings (9/2003)
- Development of a baseline design for s/c motor meeting air-worthiness targets for rating and performance (9/2005)
- Integrate design to an all-electric transport concept and to new architectures for aircraft power systems (based on fuel cell generation). Define and initiate a program to build a relevant prototype (9/2007)

## 2.5.3 Proposed Approach

- Develop database of relevant material properties (superconductor, insulation, cryogenic, etc.) for the design of high-performance s/c motors
- Establish power rating targets for aircraft-relevant motors. Develop performance targets for s/c motor (power density, weight, etc.)
- Develop motor concepts capable of attaining targets. Perform trade-off to select most promising concept. Proceed with a detailed design of the s/c motor based on the preferred concept
- Integrate findings from this task to subtasks on electrical network architecture (2.5.2) and SO fuel cell studies (2.5.1), as well as overall system integration tasks (2.1.1 and 2.1.2)
- Determine best path for a prototype development program (sub-scale or full-scale), establish goals, schedule, and budget, and proceed with hardware demonstration program as resources allow